United States of America
Department of Transportation -- Federal Aviation Administration

Supplemental Type Certificate
Number SA02717NY

This certificate issued to VoiceFlight Systems LLC
105 Jordan Road
Troy, NY 12180

certifies that the change in the type design for the following product with the limitations and conditions thereafter as specified hereon meets the airworthiness requirements of Part * of the * Regulations.

Original Product -- Type Certificate Number: *
Make: *
Model: *

* See attached FAA Approved Model List (AML). Document No. SA02717NY for the list of approved airplane models and applicable airworthiness regulations.

Description of Type Design Change:

Configuration 1 (GPS without WAAS):

Configuration 2 (GPS with WAAS):

Configuration 3 (GPS with, or without WAAS):

(See Continuation Sheet 2 of 3)

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application: November 21, 2006

Date of issuance: July 21, 2009

Date revised: November 18, 2010


By direction of the Administrator

Gaetano Sciortino
Manager
New York Aircraft Certification Office

Signature

Any alteration of this certificate is punishable by a fine of not exceeding $1,000, or imprisonment not exceeding 3 years, or both.

FAA Form 8110-2(10-68) Page 1 of 3

This certificate may be transferred in accordance with FAR 21.47.
Limitations and Conditions:

Configuration 1:
3. The VFS101 must interface with an existing Garmin GNS 430 or Garmin GNS 530. The installation of the Garmin 430/530 is not approved by this STC.
4. Acceptable GPS installations are limited to a single Garmin 430, dual Garmin 430, single Garmin 530, dual Garmin 530, or a single Garmin 430 plus a single Garmin 530.
5. The VFS101 must interface with an audio panel with an available un-switched input having a nominal input impedance of no less than 500 Ohms. The installation of the audio panel is not approved by this STC.
6. Installation not approved for aircraft with a Maximum Certificated Gross Takeoff Weight of 6,000 pounds or more.
7. The installer must determine whether this design change is compatible with previously approved modifications.
8. If the holder agrees to permit another person to use this certificate to alter a product, the holder must give the other person written evidence of that permission.

Configuration 2:
3. The VFS101 must interface with an existing Garmin GNS 430W or Garmin GNS 530W. The installation of the Garmin 430W/530W is not approved by this STC.
4. Acceptable GPS installations are limited to a single Garmin 430W, dual Garmin 430W, single Garmin 530W, dual Garmin 530W, or a single Garmin 430W plus a single Garmin 530W.
5. The VFS101 must interface with an audio panel with an available un-switched input having a nominal input impedance of no less than 500 Ohms. The installation of the audio panel is not approved by this STC.
6. Installation not approved for aircraft with a Maximum Certificated Gross Takeoff Weight of 6,000 pounds or more.
7. The installer must determine whether this design change is compatible with previously approved modifications.
8. If the holder agrees to permit another person to use this certificate to alter a product, the holder must give the other person written evidence of that permission.

Configuration 3:
3. The VFS101 must interface with one or two existing Garmin 400/500 Series (Non-WAAS) GPS units, or with one or two existing Garmin 400W/500W Series (WAAS) GPS units. The installation of the Garmin GPS units is not approved by this STC.
4. The VFS101 must interface with an audio panel with an available un-switched input having a nominal input impedance of no less than 500 Ohms. The installation of the audio panel is not approved by this STC.
5. Installation not approved for aircraft with a Maximum Certificated Gross Takeoff Weight of 6,000 pounds or more.
6. The installer must determine whether this design change is compatible with previously approved modifications.
7. If the holder agrees to permit another person to use this certificate to alter a product, the holder must give the other person written evidence of that permission.

(See Continuation Sheet 3 of 3)
Supplemental Type Certificate
(Continuation Sheet)

Number SA02717NY
Date Amended: January 30, 2013

Certification Basis:

The certification basis for the changed aircraft as described in this certificate is as follows:

a. The type certification basis as shown on the TCDS (see AML, Document No. SA02717NY) for all areas not affected by the change.

b. The certification basis for parts changed or affected by the change since the reference date of application, November 21, 2006, is as shown on the TCDS (see AML, Document No. SA02717NY). In addition, the holder of this certificate has complied with 14 CFR Part 23 as amended by Amendment 23-54.

The following lists the Federal Aviation Regulations complied with through Amendment Level 2354:

|-------|-------|-------|-------|-------|-------|-------|-------|-------|-------|

---END---